

# Numerical Simulation of the Downstream Fan Noise of a Coaxial Jet with a Shielding Surface

*SEBU project (SEmi-BURied engine), co-funded by Airbus SAS and  
EREA*

**Stéphane REDONNET, Eric MANOHA**  
**ONERA (French Aerospace Centre)**  
***Department of Computational Fluid Dynamics and Aeroacoustics,***

***and***

**Owen KENNING**  
***QinetiQ (Aerospace Centre)***  
***Air Systems Technology Department***

*“Aeroacoustics of New Aircraft & Engine Configurations”, 11<sup>th</sup> -12<sup>th</sup> November 2004, Budapest  
(Hungary)*

# Objective

*Isolated nozzle*

*Shielded nozzle*

**Comparative study of both a classical and a modified co-axial engine, in order to evaluate both the :**

- **installation shielding effect (secondary exhaust extension)**
- **highly inhomogeneous mean flow effect**

**on**

**the downstream rear tone fan noise**

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## General methodology

**Preliminary RANS (3D/ 2D-axi) mean flow computation : CFD solver**

$V_0$   
→

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**Near & mid-field propagation: n:  
Euler's solver**

$V_0$   
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**Preliminary RANS (3D/ 2D-axi) mean flow computation : CFD solver**

$V_0$   
→



Kirchhoff frontier

Kirchhoff frontier

**Far field radiation :  
Kirchhoff's solver**

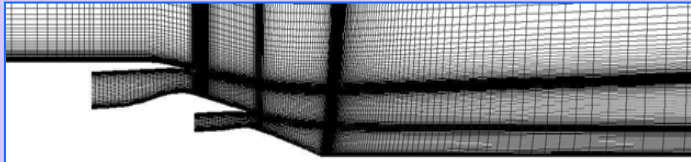
**Near & mid-field propagation: n:  
Euler's solver**

# Outline

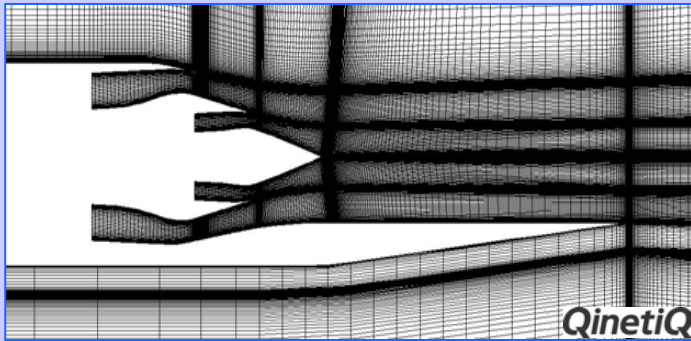
- ❑ Preliminary tasks
- ❑ CAA Hybrid Methodology Tools
- ❑ 2D study of the Shielding Effect in a Quiescent Medium
- ❑ 2D study of the Shielding Effect in an Inhomogeneous Medium
- ❑ Conclusions of the 2D studies : Shielding and Mean Flow Effects
- ❑ Early 3D results : Isolated Nozzle in both a Quiescent and an Inhomogeneous Medium
- ❑ Conclusions & Perspectives

# Preliminary tasks

# Preliminary CFD (RANS) Computations (*QinetiQ*, *Fluent v.5*)

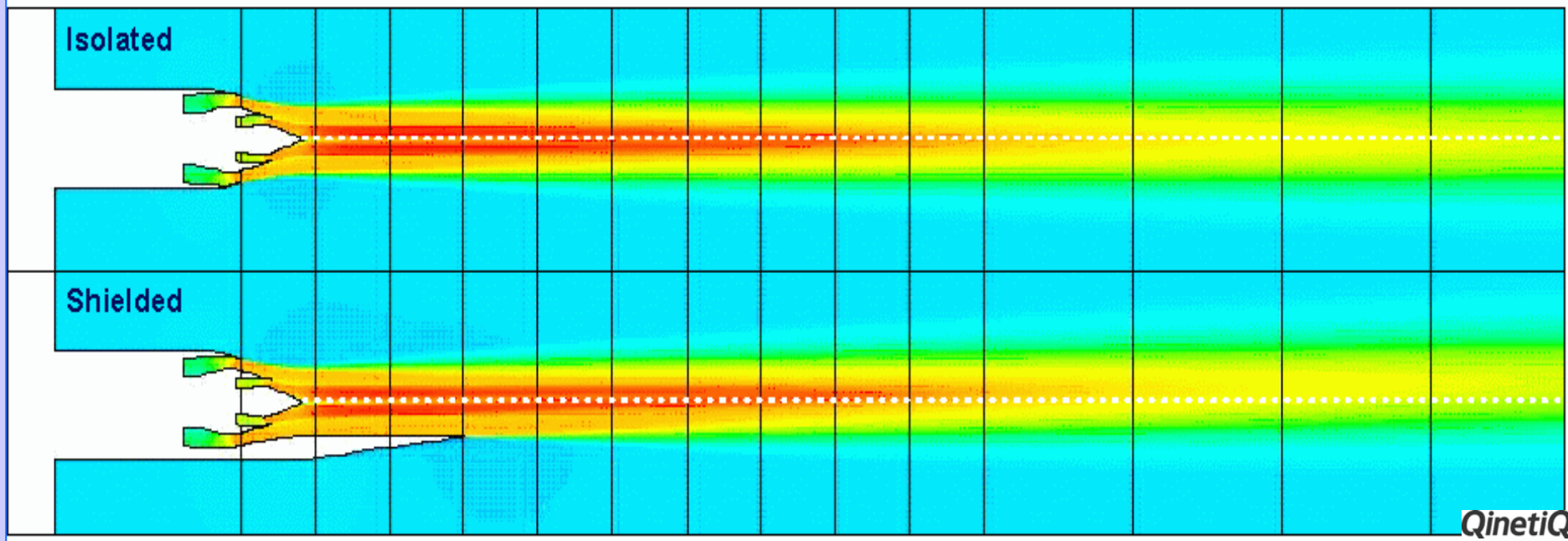


*Isolated axi-symmetric nozzle (2D mesh, partial view):  
5 structured domains (94 805 elements)*



*Shielded nozzle (3D mesh, partial view):  
10 structured domains (5 542 100 elements)*

*High Subsonic Hot Jet, Take Off Conditions ( $M^\infty = 0.25$  /  $M^{max} \approx 0.9$ ) : Axial velocity results*





# CAA Hybrid Methodology Tools

## Near & Mid-Fields Propagation over Inhomogeneous Flows & Solid Bodies : *sAbrinA*

*("Solver for Acoustic BRoadband Interaction with Aerodynamics")*

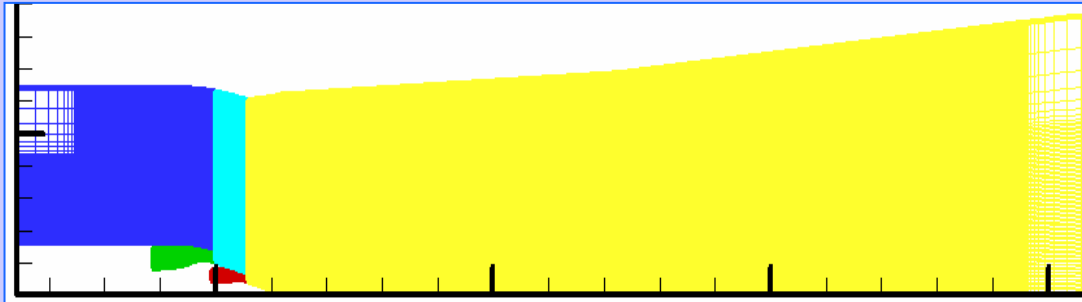
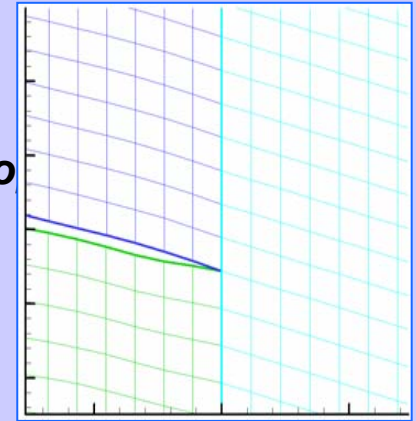
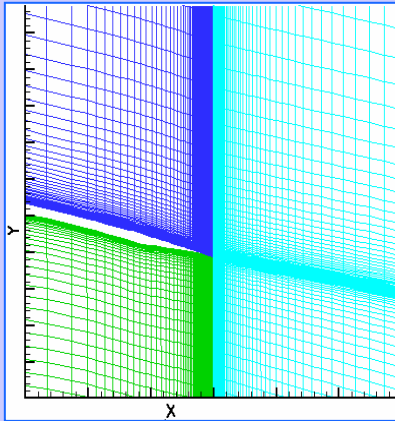
- \* Full Euler or Navier-Stokes equations
- \* Complete (NLDE) or Splitted variables (hybrid CAA)
- \* Finite Differences or Volumes
- \* High-order (up to 6<sup>th</sup> order in space, up to 3<sup>rd</sup> order in time, up to 10<sup>th</sup> order for filters)
- \* Multi-D, multi-domain
- \* Specific boundary conditions (rigid obstacle, symmetry plane, free-field)

## Far Field Radiation in Homogeneous Medium : *Kirch2D*, *Kirch3D*

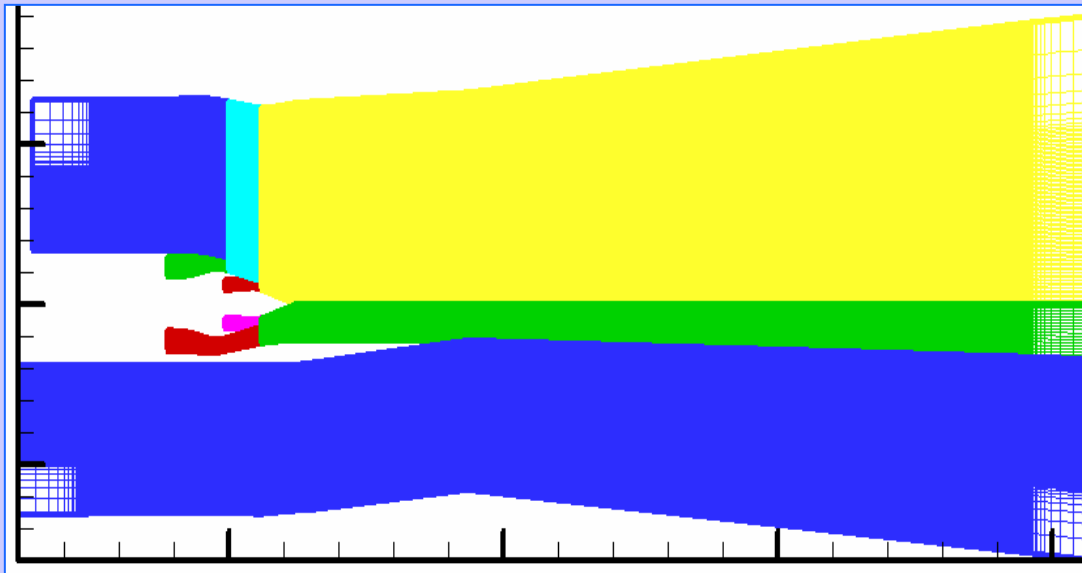
- \* Frequential 2D or 3D codes (Kirchhoff integration)
- \* Frequency or temporal, harmonic or broadband signals
- \* Homogeneous medium (at rest or with uniform flow)

# 2D Acoustic Grid Derivation

- . Conservation of blocks number, frontiers and topology*
- . Complete re-meshing of each block*



*Isolated nozzle (2D mesh):  
5 structured domains (64 408 elements)*

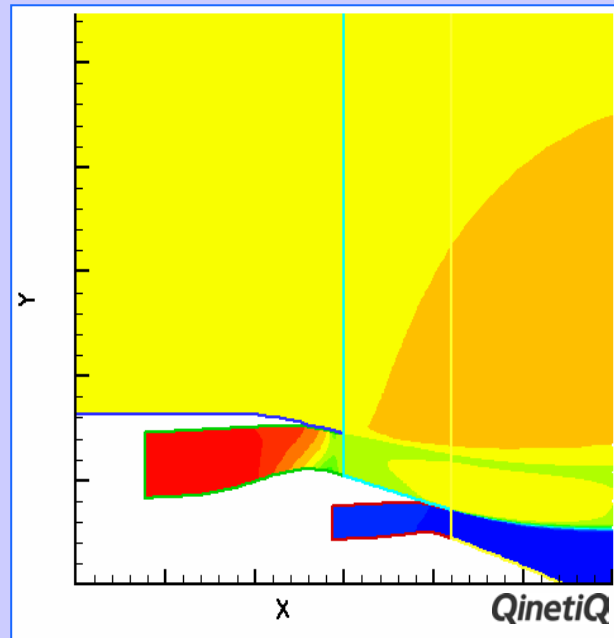


*Shielded nozzle (2D mesh):  
9 structured domains (154 442 elements)*

$$\Delta y_{min\ CAA} = 500 \Delta y_{min\ CFD} = \lambda/10$$

# 2D RANS Mean Flow Interpolation (*Barycentral bilinear technique*)

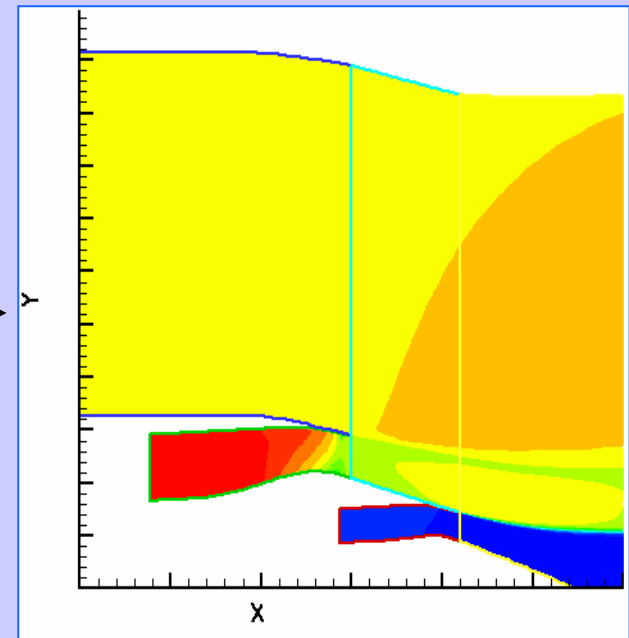
*Density field on the CFD grid*



*Barycentral  
bilinear  
technique*



*Density field on the CAA grid*



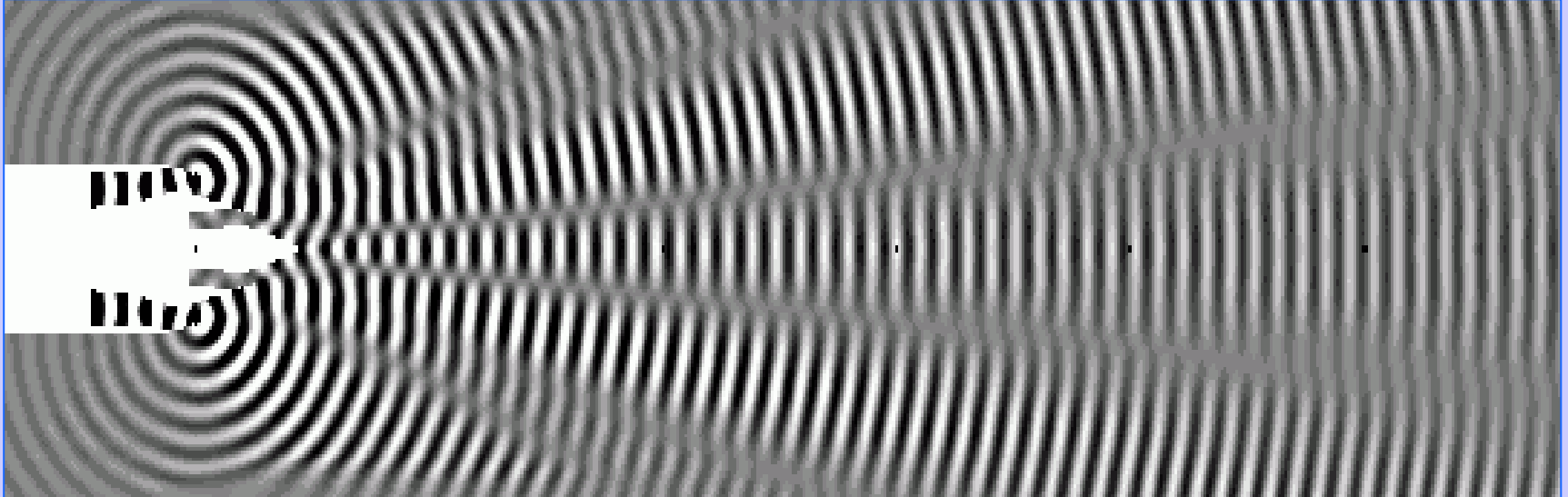
# **Shielding Effect on the Fan Noise 2D Propagation in a Quiescent Medium**

# Fan Noise in a Quiescent Medium

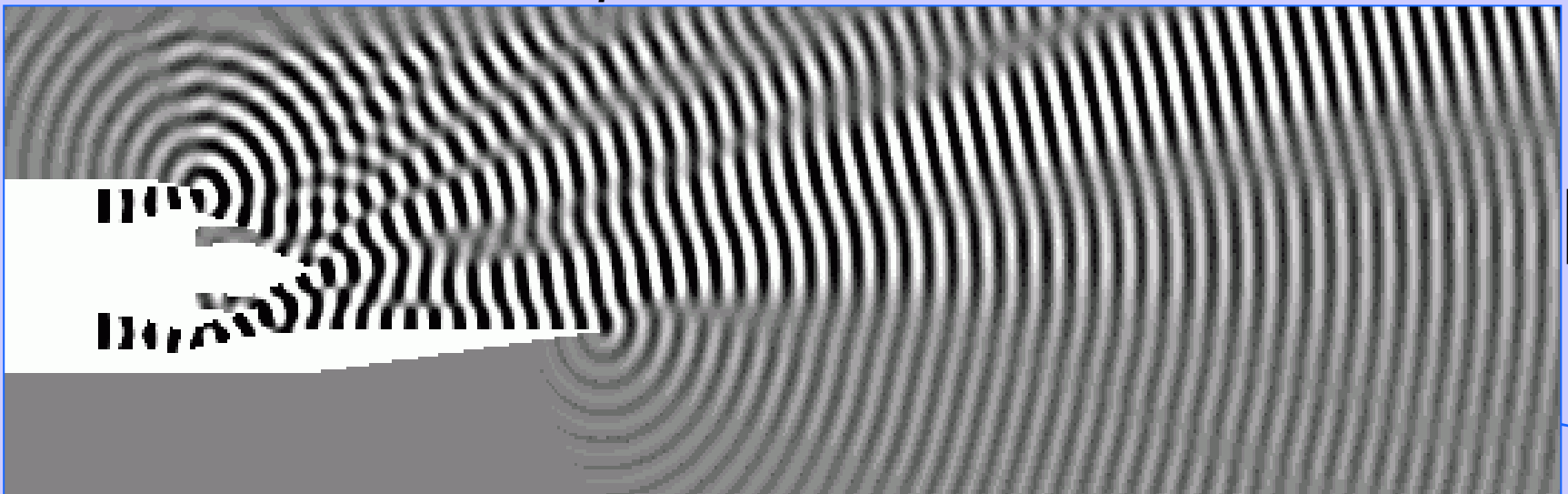
*(1/3) Near & Mid-Fields 2D Propagation (sAbrinA code)*

*Isolated nozzle : Instantaneous pressure field*

$kR = 20$



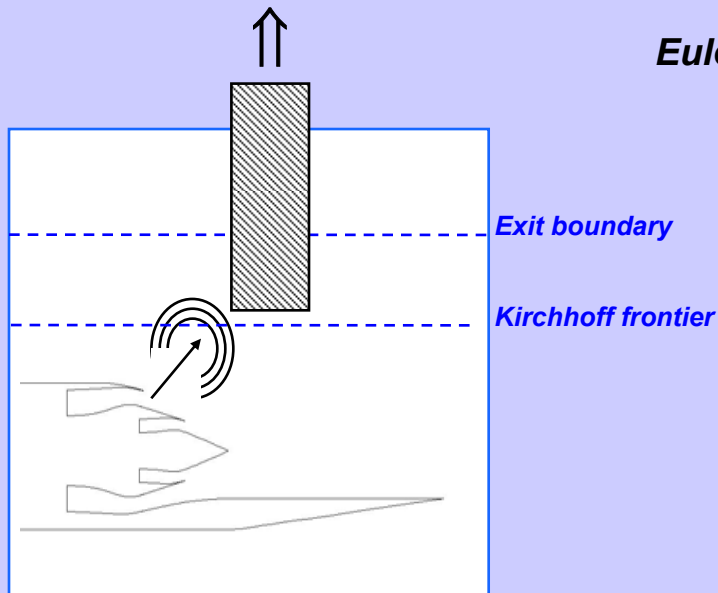
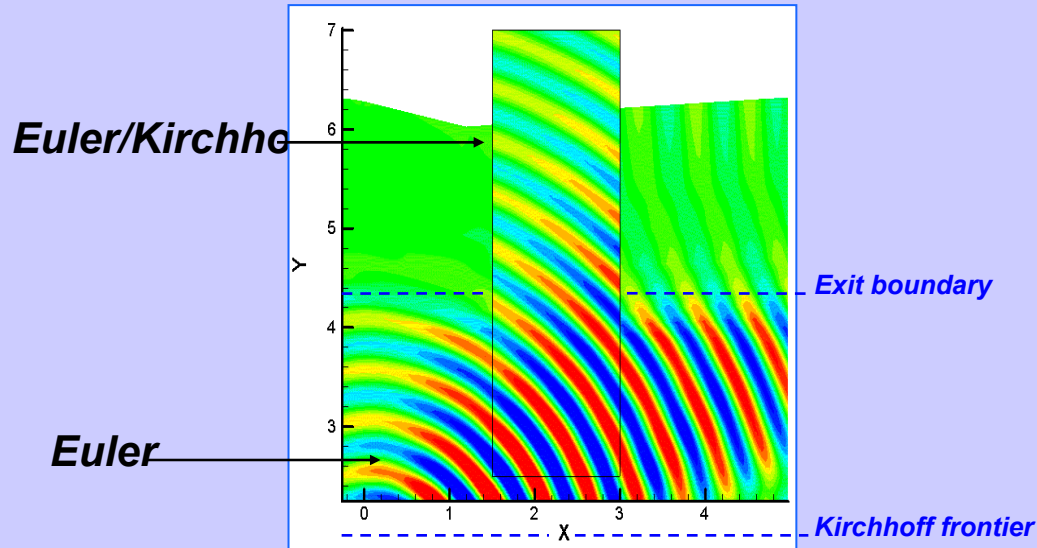
*Shielded nozzle: Instantaneous pressure field*



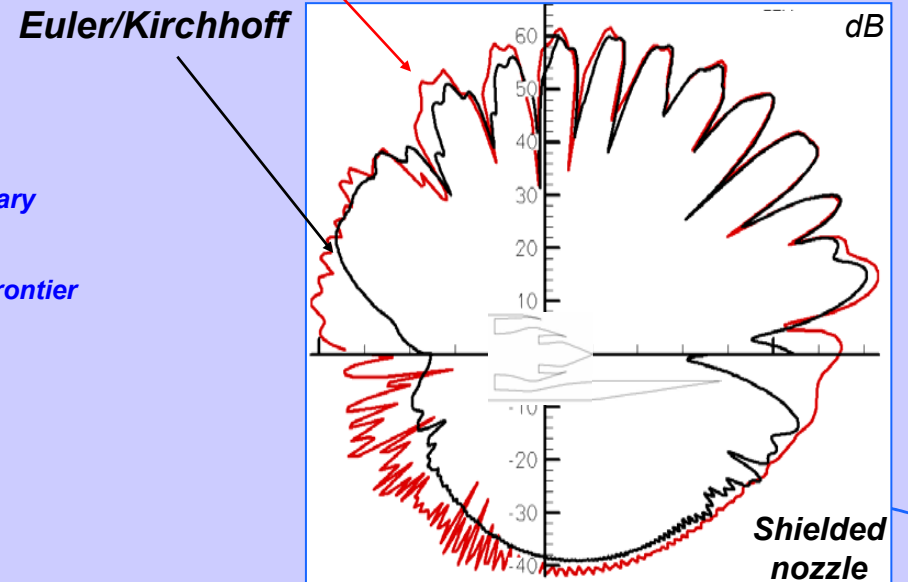
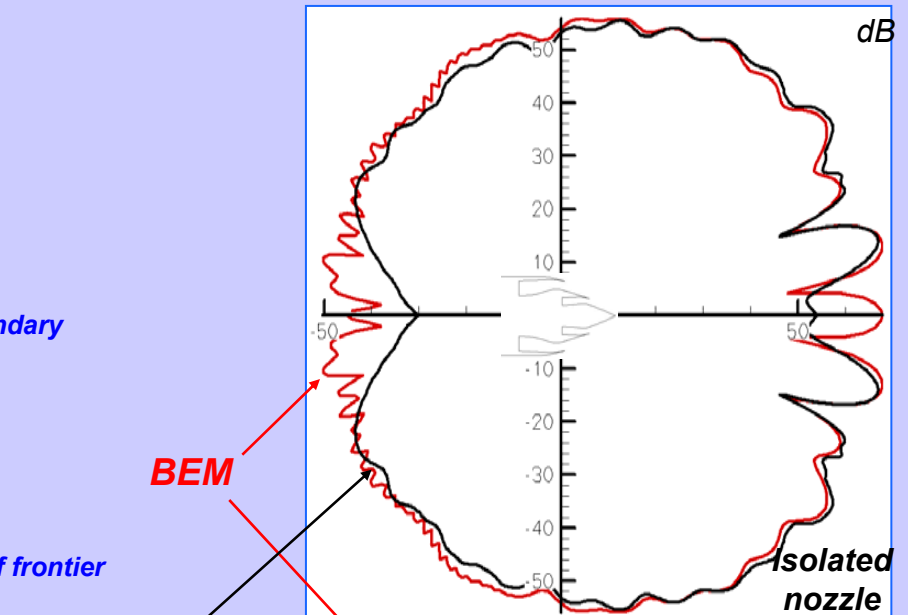
# Fan Noise in a Quiescent Medium

## (2/3) Far-Field 2D Radiation (Kirch2D code) and Validation (Sysnoise code)

**Kirch2D validation**



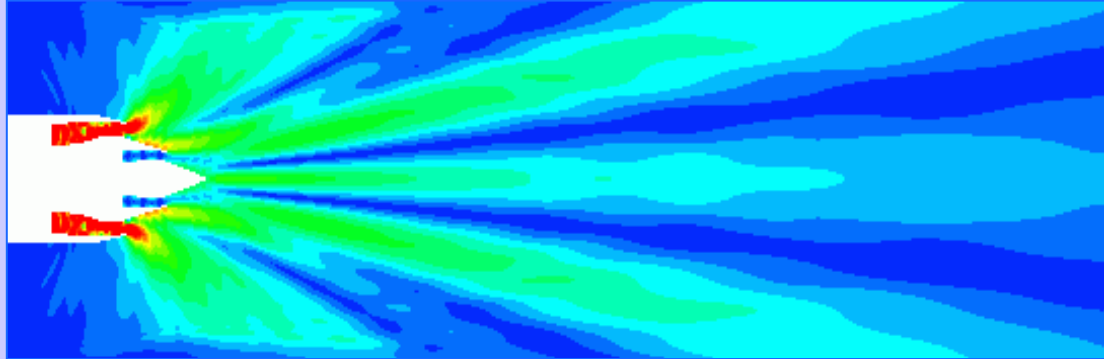
**sAbrinA/Kirch2D validation**



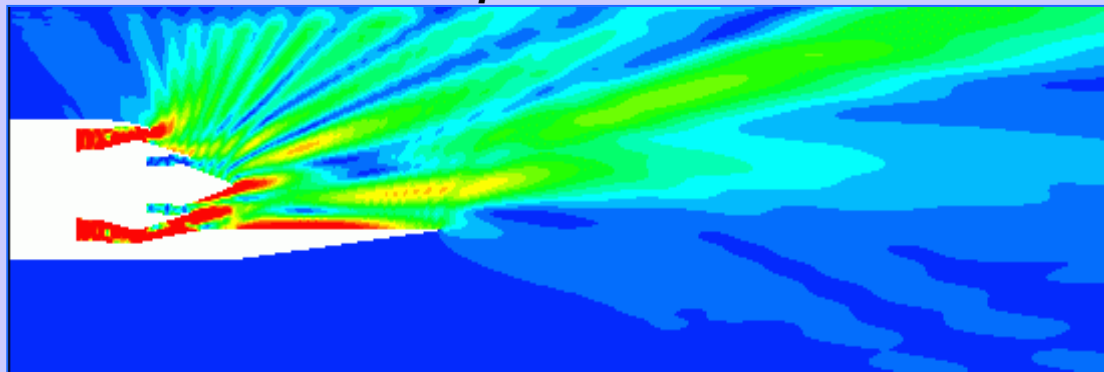
# Fan Noise in a Quiescent Medium

## (3/3) 2D Shielding Effect Analysis

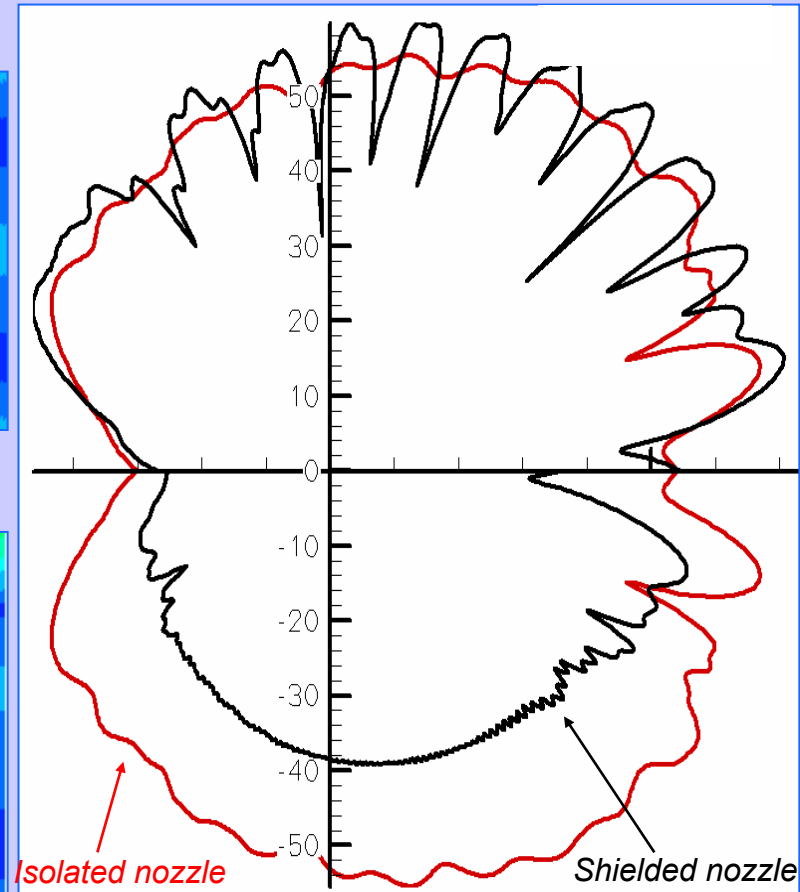
**Isolated nozzle : RMS pressure near & mid-fields**



**Shielded nozzle : RMS pressure near & mid-fields**



**Far field directivity diagrams (dB)**



# **Shielding Effect on the Fan Noise 2D Propagation in an Inhomogeneous Medium**

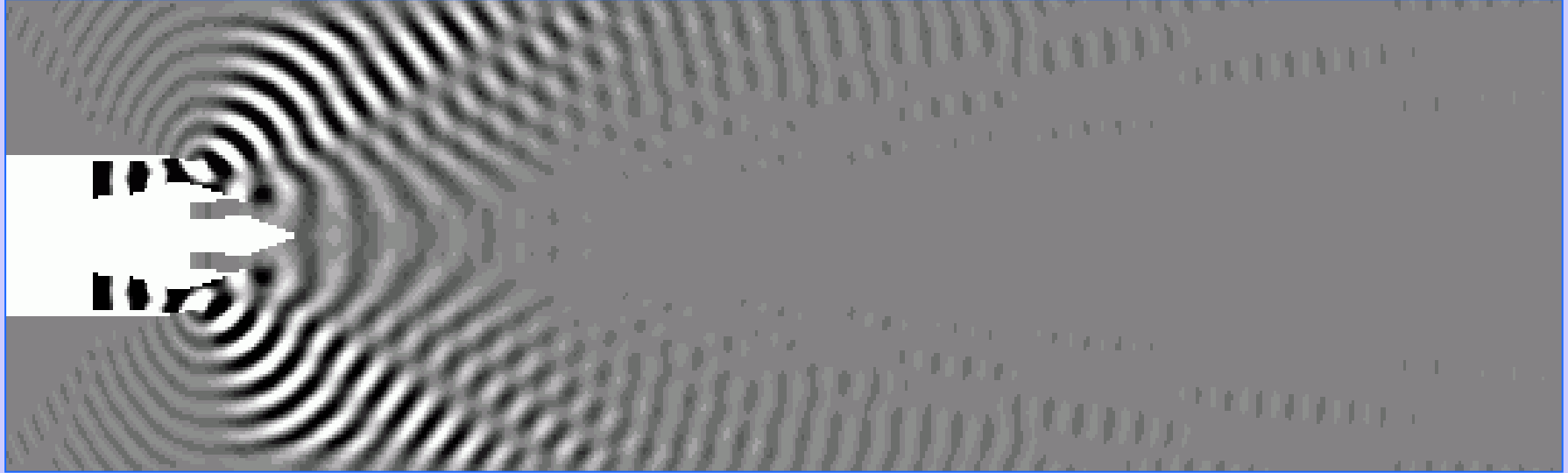


# Fan Noise in an Inhomogeneous Medium

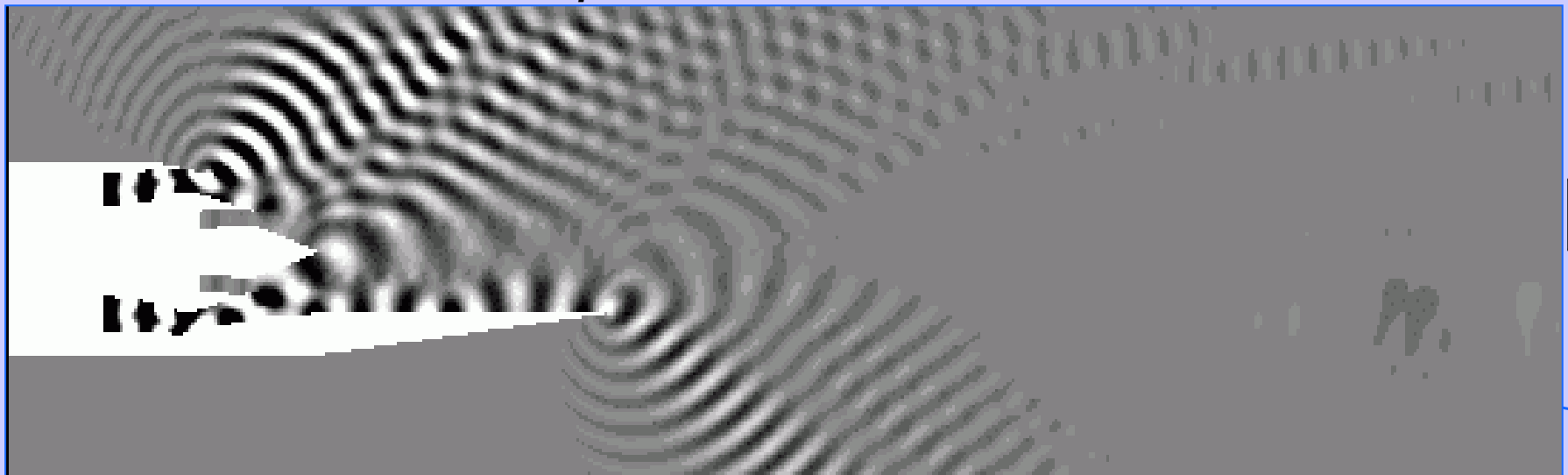
## (1/3) Near & Mid-Field 2D Propagation (sAbrinA code)

*Isolated nozzle : Instantaneous pressure field*

$kR = 20$



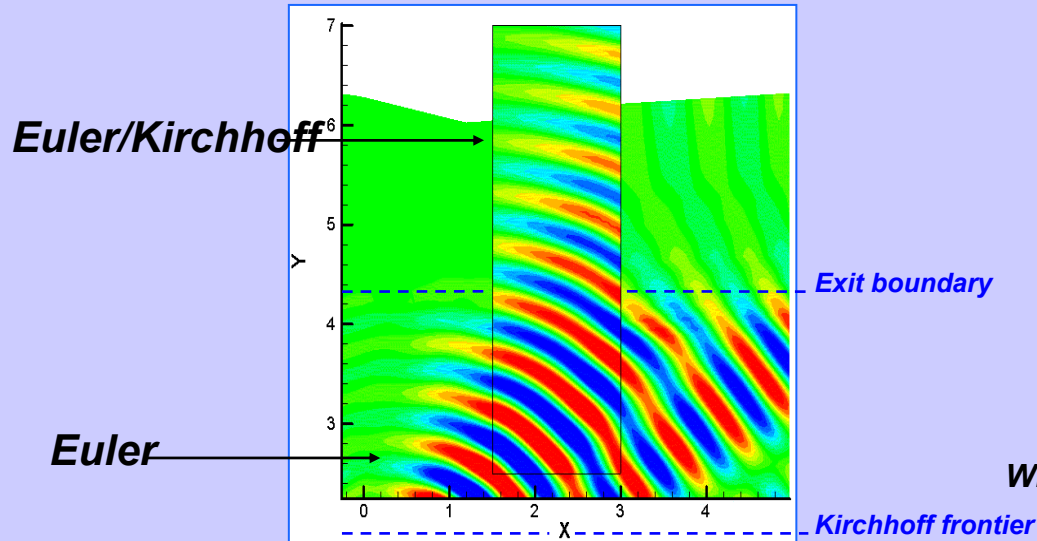
*Shielded nozzle: Instantaneous pressure field*



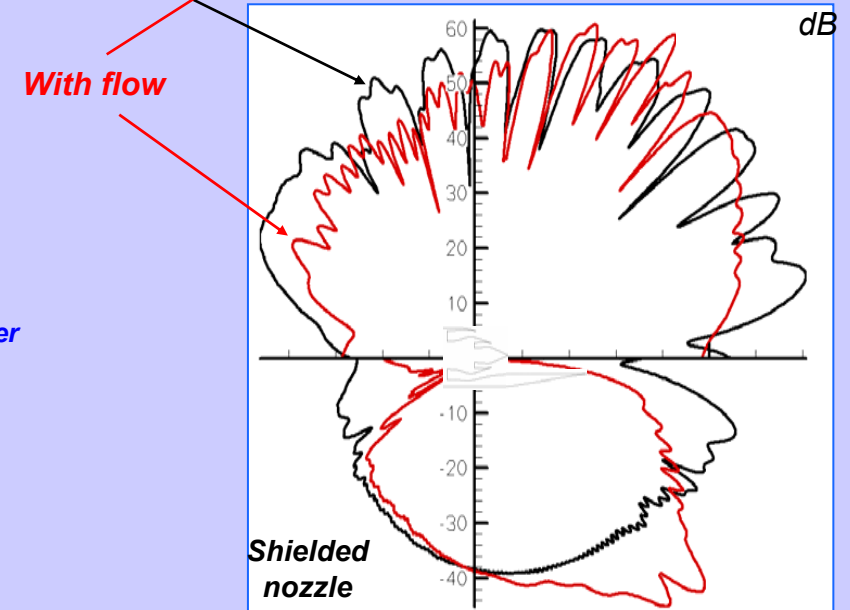
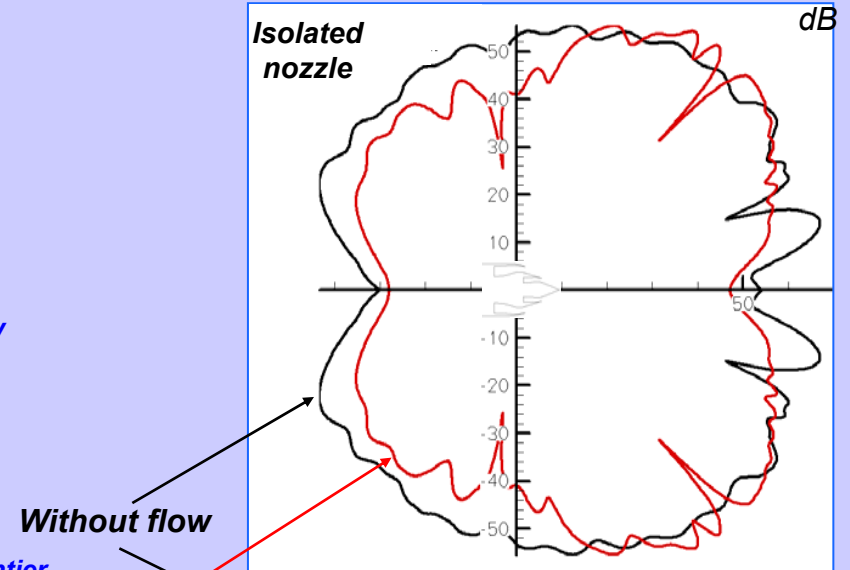
# Fan Noise in an Inhomogeneous Medium

(2/3) Far-Field 2D Radiation (Kirch2D code)

**Kirch2D validation**



**sAbrinA/ Kirch2D validation**



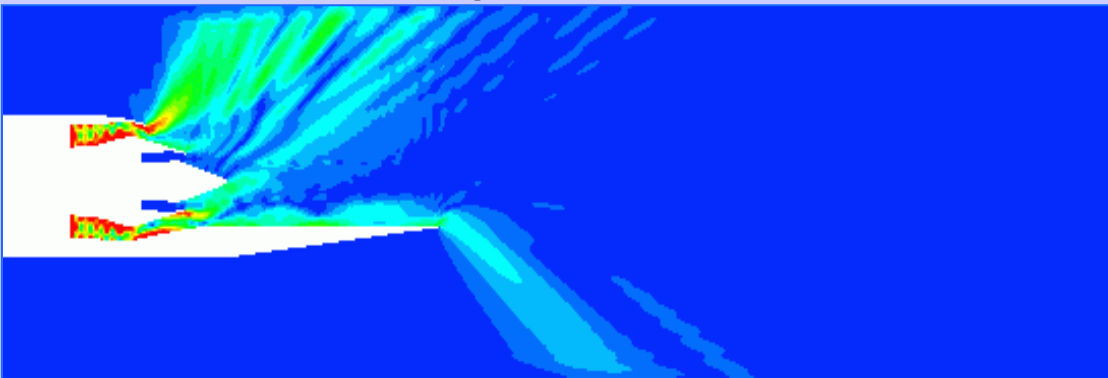
# Fan Noise in an Inhomogeneous Medium

## (3/3) 2D Shielding Effect Analysis

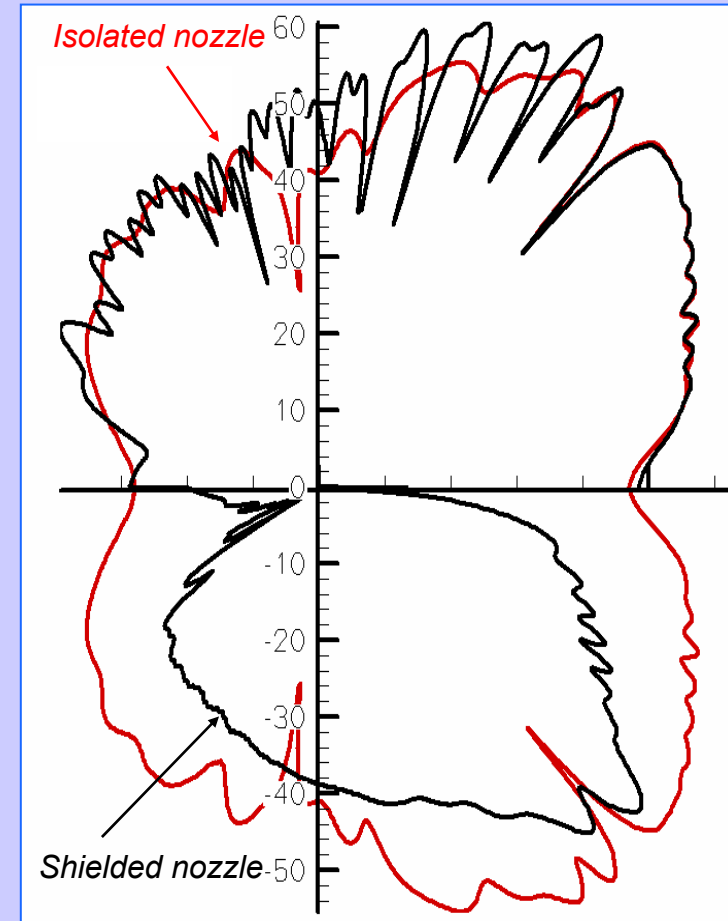
**Isolated nozzle : RMS pressure near & mid-fields**



**Shielded nozzle : RMS pressure near & mid-fields**



**Far field directivity diagrams (dB)**



# Fan Noise : Conclusions

## *Shielding & Highly Sheared Mean Flow 2D Effects Analysis*

$kR = 20$

*Isolated nozzle, without flow*

*Isolated nozzle, with flow*

Mean flow  
↔  
effect

Shielding  
↕  
effect

Mean flow  
↔  
effect

*Shielded nozzle, with flow*

Shielding  
↕  
effect

*Shielded nozzle, without flow*

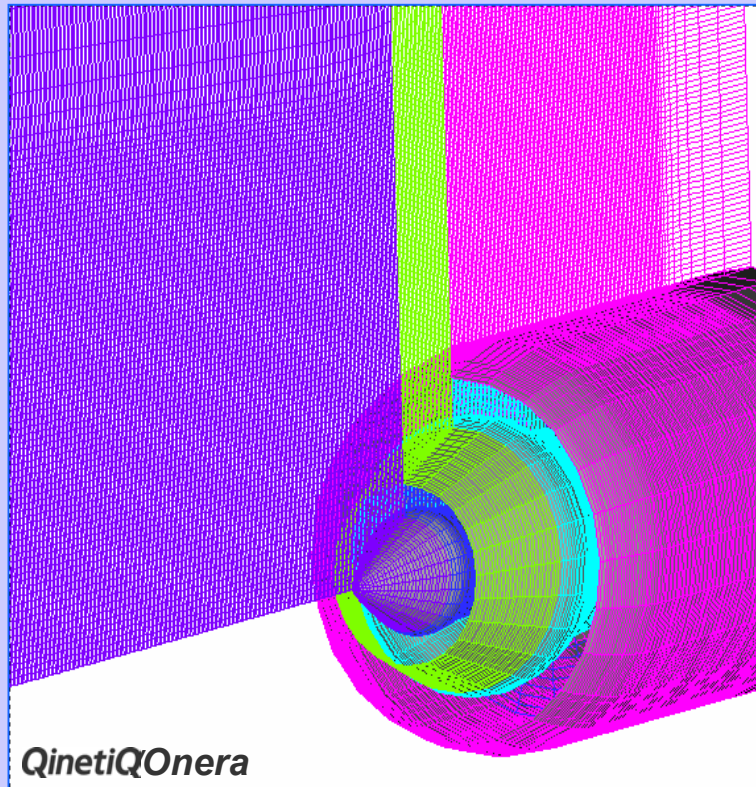
- ***Important shielding effect on the rear fan noise in a quiescent medium***
  - ***Important mean flow refraction effect on the rear fan noise***
- ***Significant attenuation of the shielding effect by the mean flow***

# **Fan Noise 3D Propagation in both a Quiescent and an Inhomogenous Medium, for the Isolated Nozzle**

# Fan Noise for the Isolated Nozzle

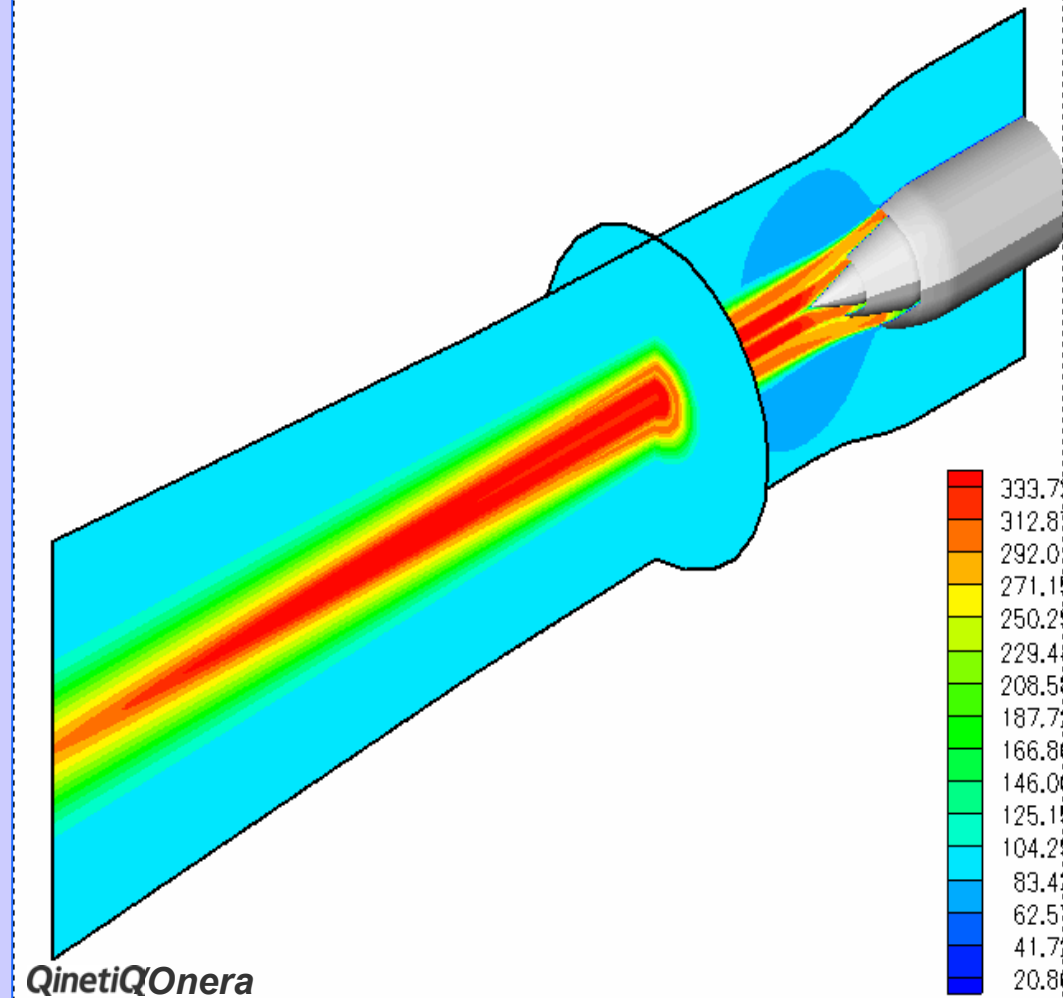
*Preliminary task : mesh and mean-flow 2D  $\Rightarrow$  3D extension*

**3D CAA mesh**  
(1,606,000 points / 25 angular planes )



**Requirements :**  
12 points per apparent angular wavelenght  
 $\Downarrow$   
spinning modes of order up to 2

**3D axi-symmetric mean flow (axial velocity)**



ONERA

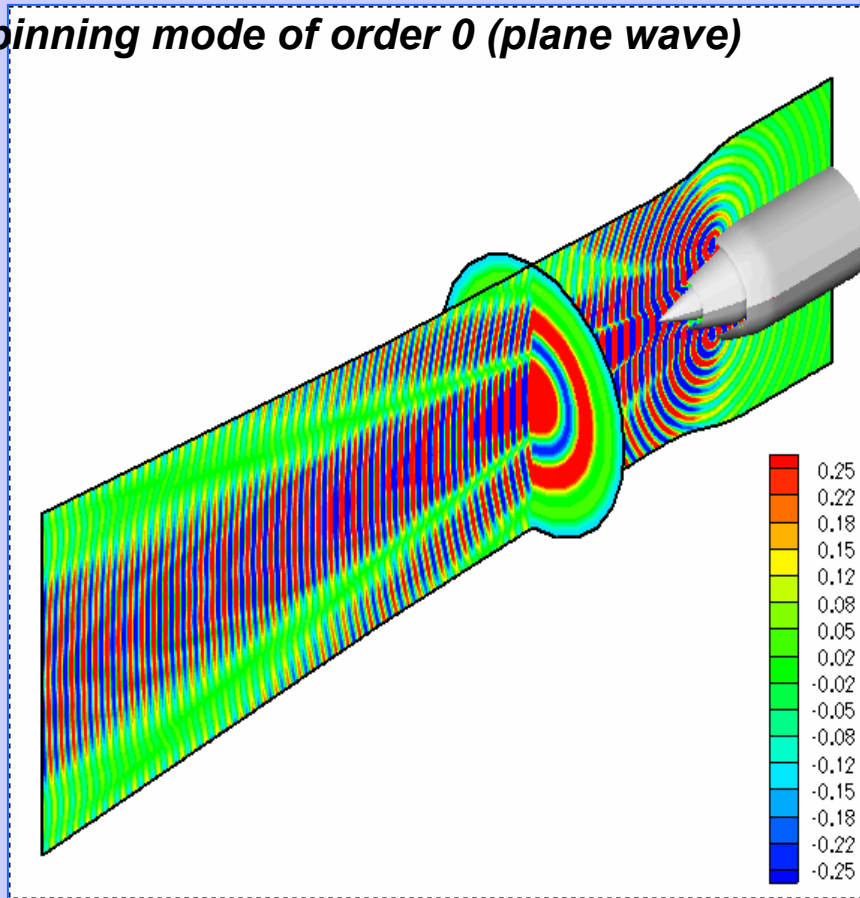


# Fan Noise in a Quiescent Medium (Isolated Nozzle)

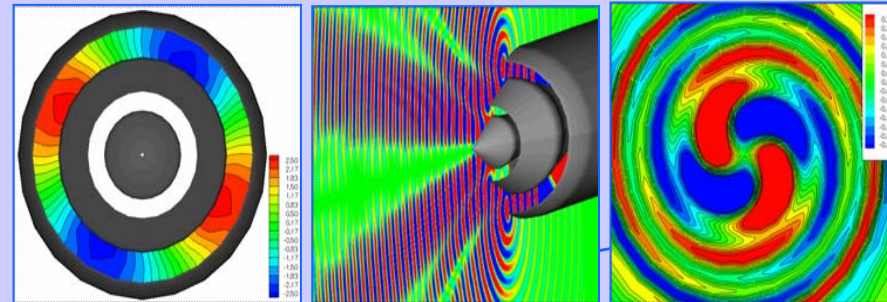
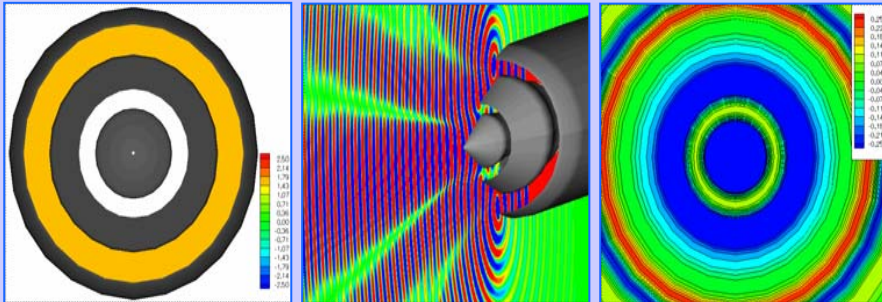
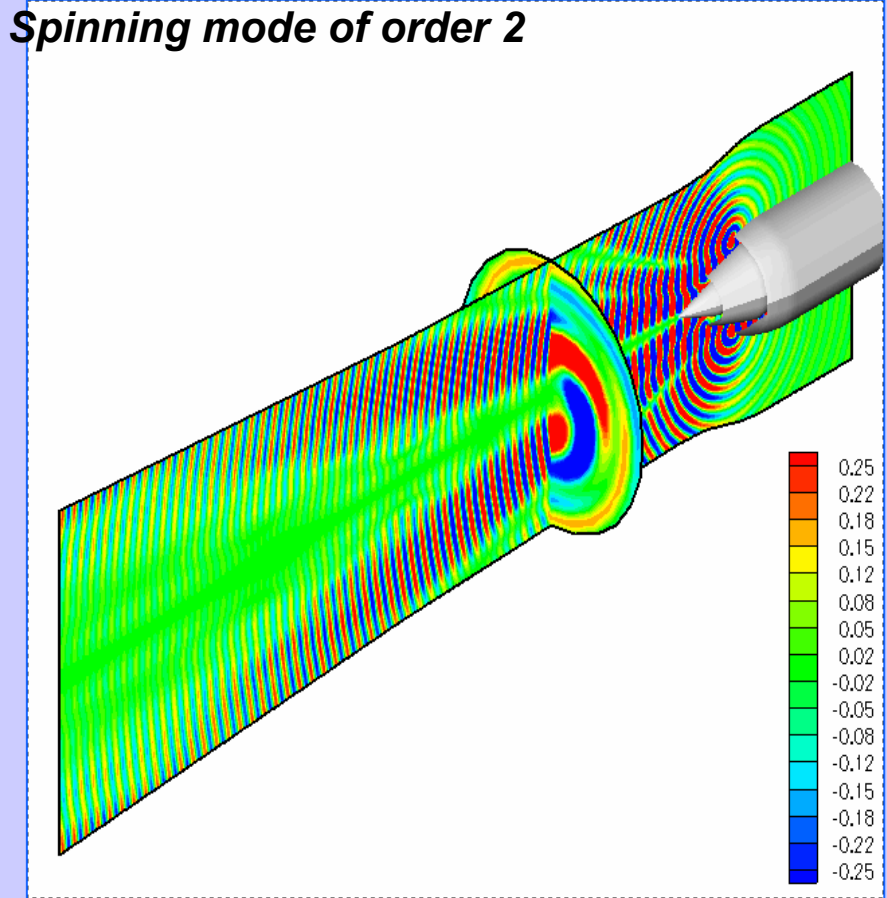
## (1/3) Near & Mid-Field 3D Propagation (sAbrinA 3D)



*Spinning mode of order 0 (plane wave)*



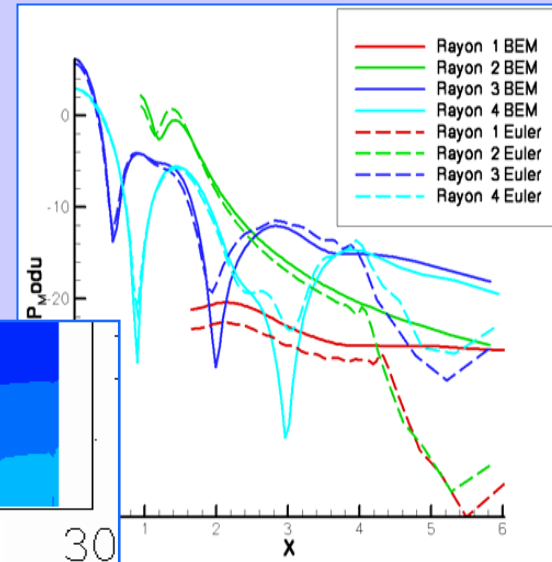
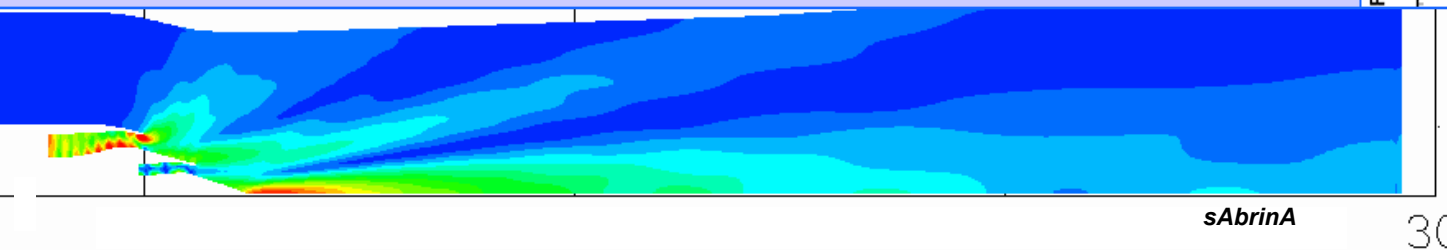
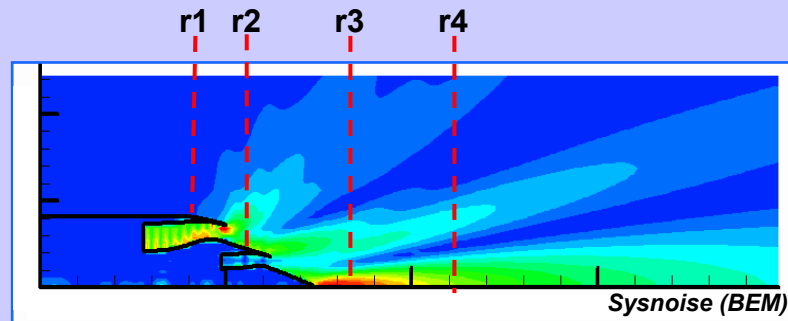
*Spinning mode of order 2*



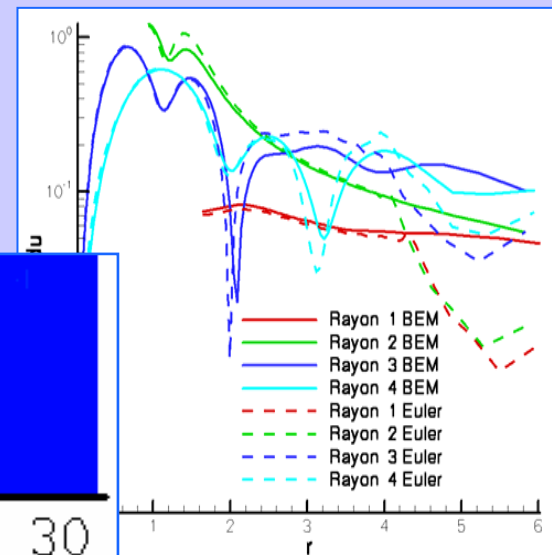
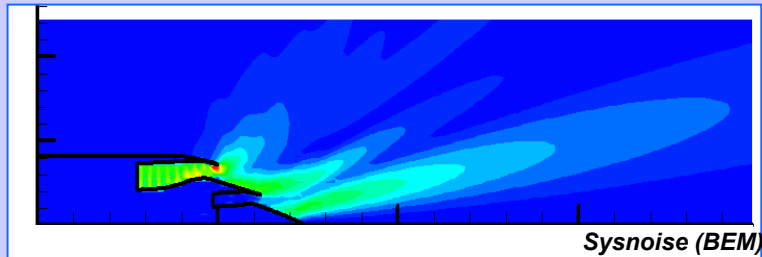
# Fan Noise in a Quiescent Medium (Isolated Nozzle)

## (2/3) Near & Mid-Field Validation (sAbrinA 3D vs. Sysnoise)

**Spinning mode of order 0  
(plane wave)**



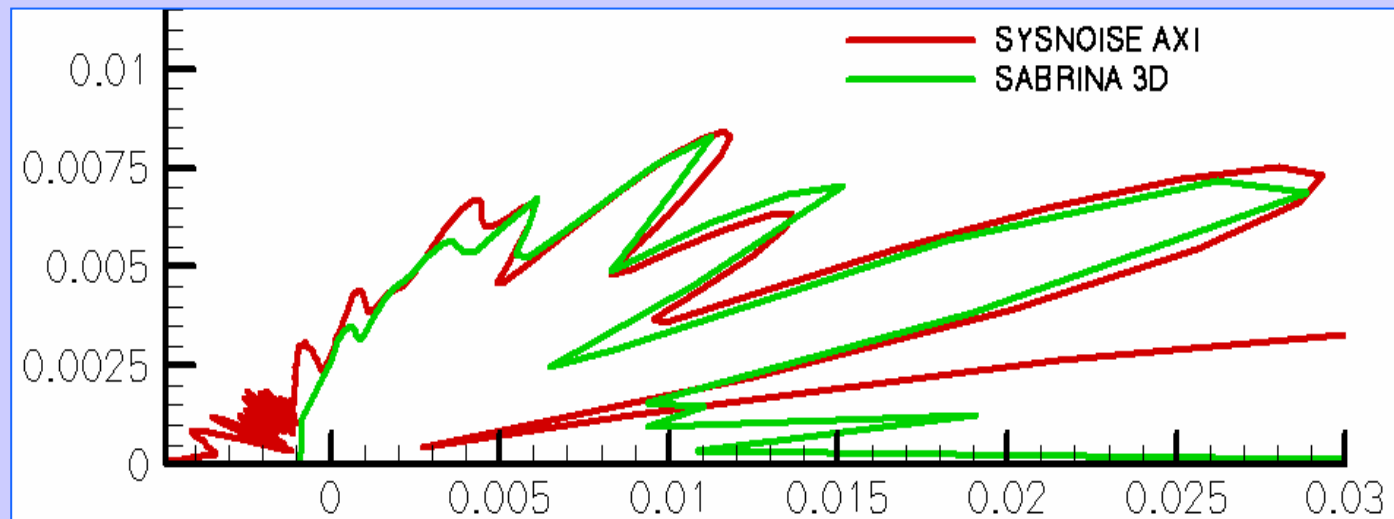
**Spinning mode of order 2**



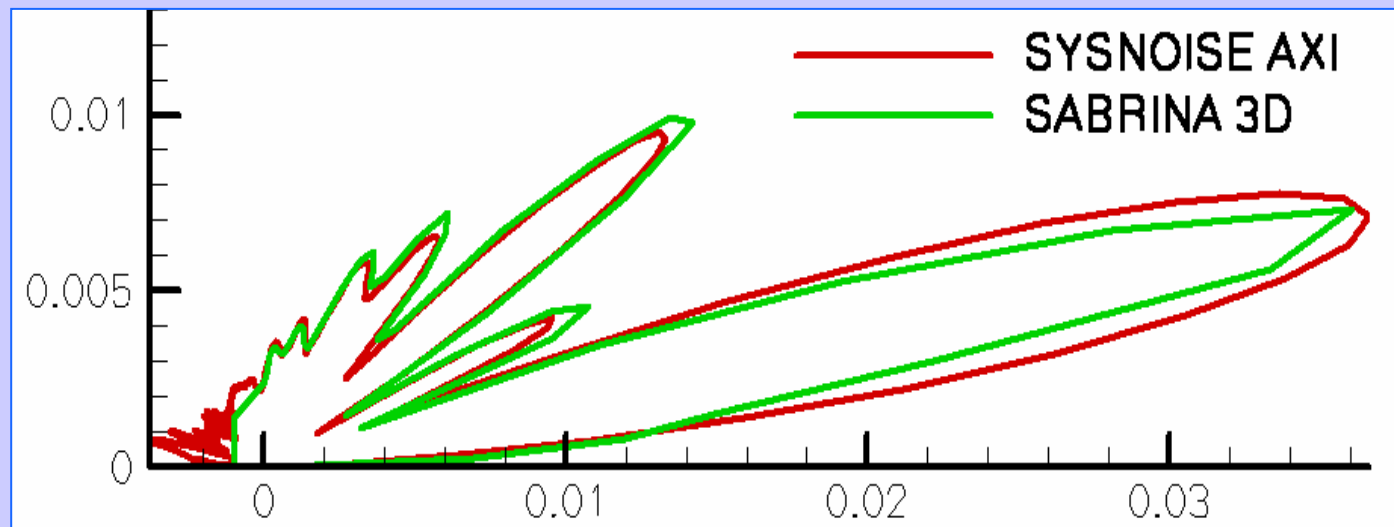


# Fan Noise in a Quiescent Medium (Isolated Nozzle)

## 3/3) Far-Field 3D Radiation & Validation (sAbrinA 3D / Kirch3D vs. Sysnoise)



*Spinning mode of order 0 (plane wave)*



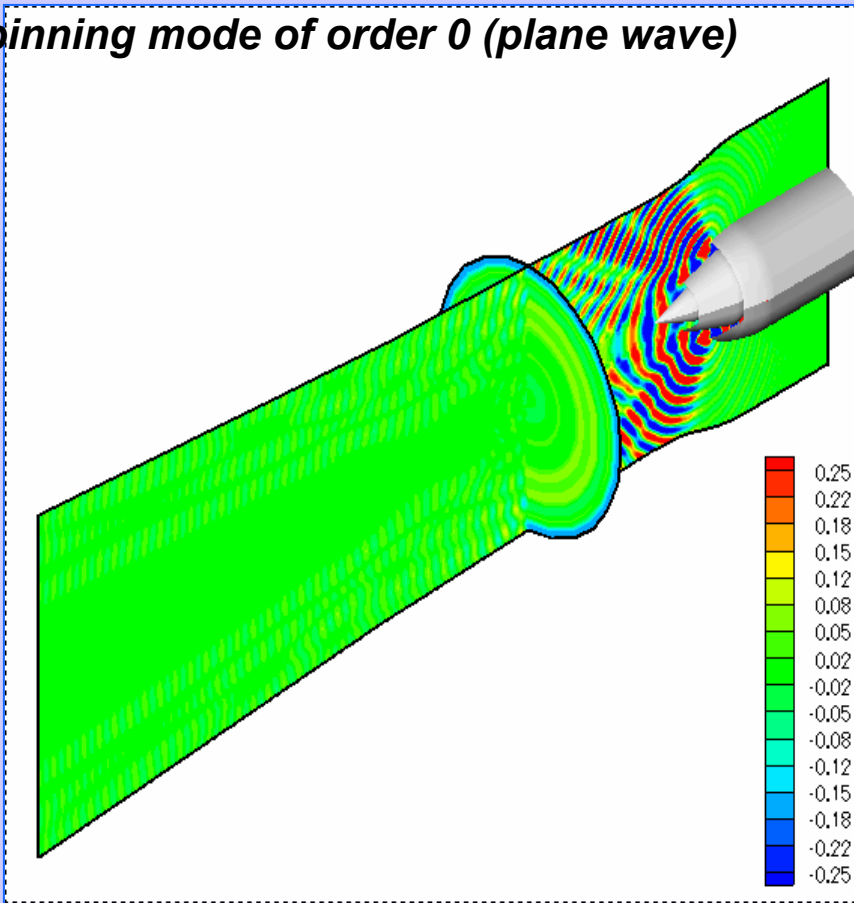
*Spinning mode of order 2*

# Fan Noise in an Inhomogeneous Medium (Isolated Nozzle)

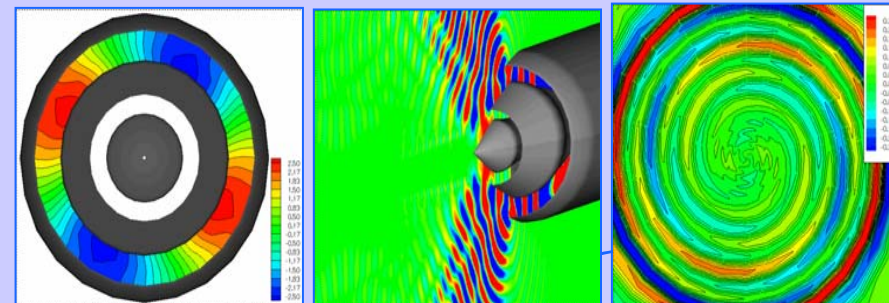
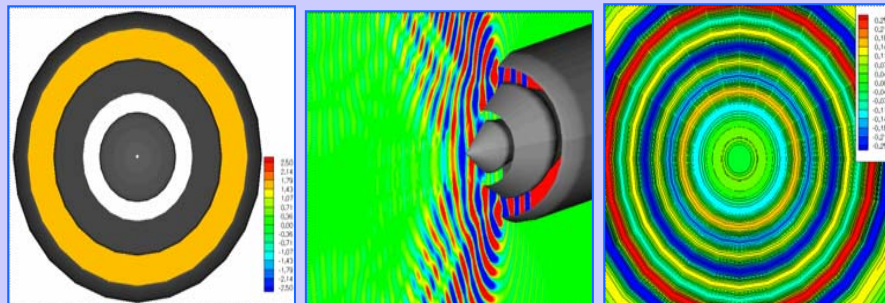
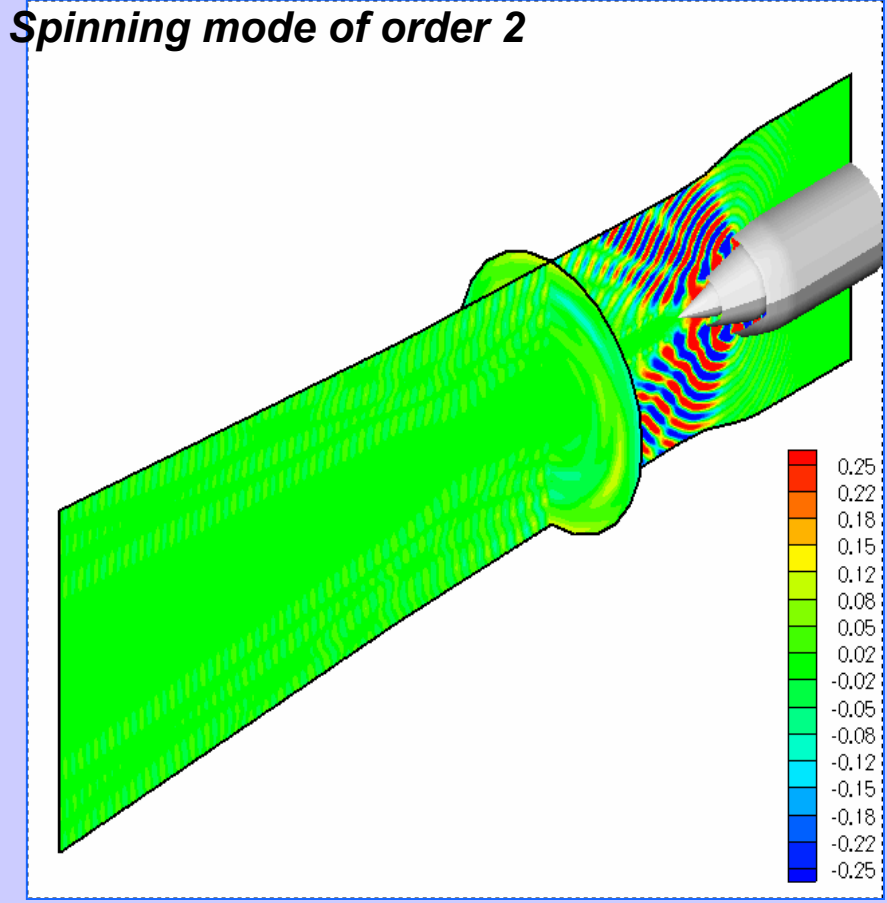
## (1/2) Near & Mid-Field 3D Propagation (sAbrinA 3D)



*Spinning mode of order 0 (plane wave)*

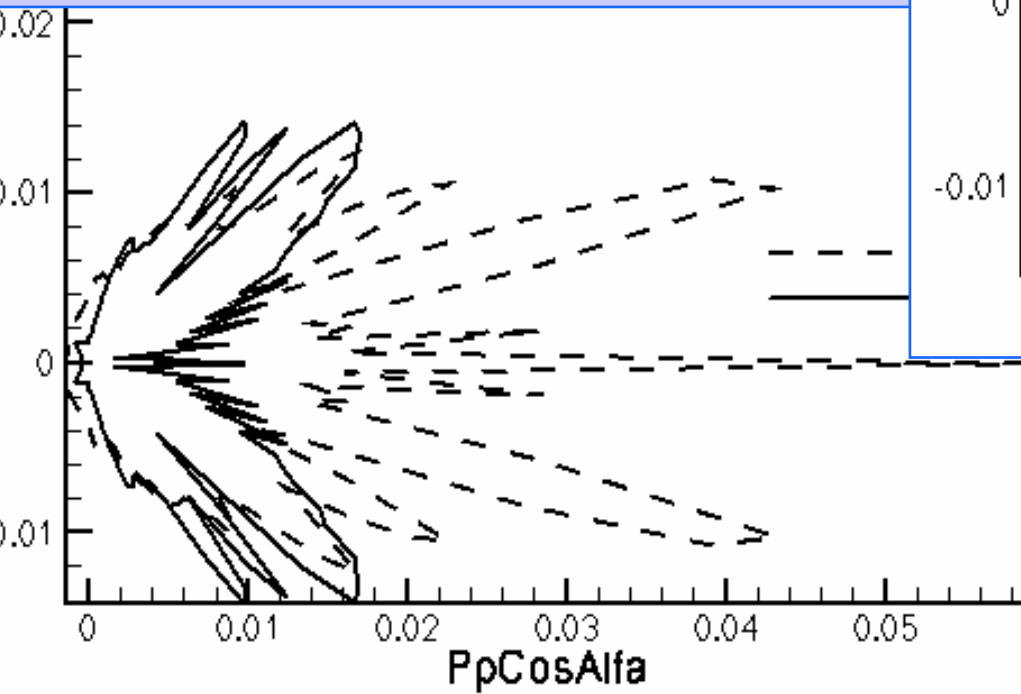


*Spinning mode of order 2*

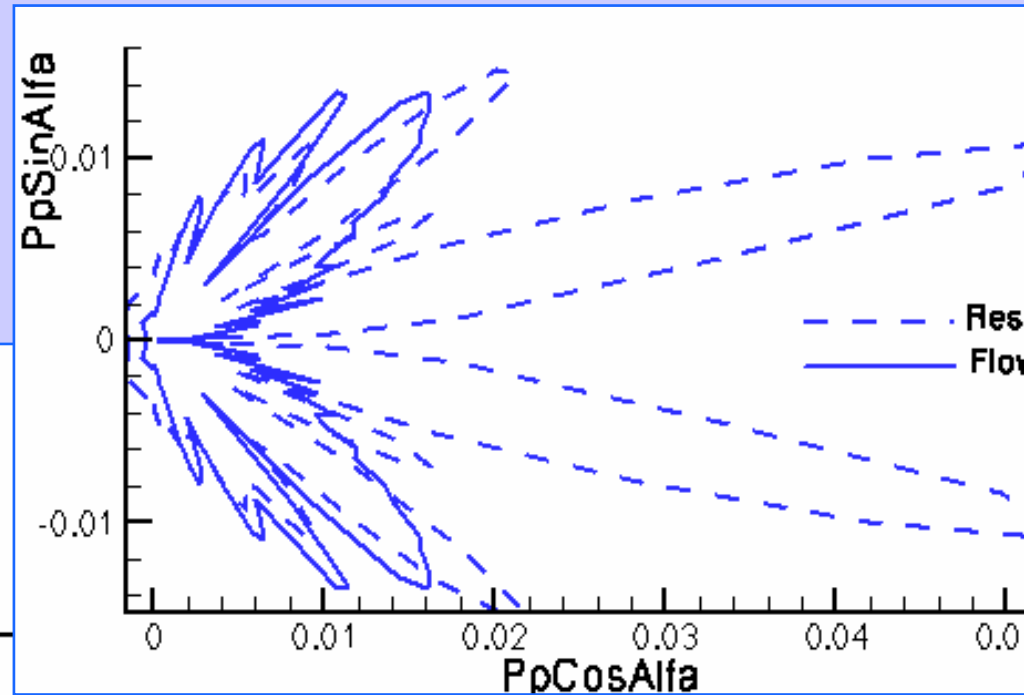


# Fan Noise in an Inhomogeneous Medium (Isolated Nozzle)

Far-Field 3D Radiation (sAbrinA 3D/ Kirch3D), and Mean Flow Effects Analysis



*Spinning mode of order 0 (plane wave)*



*Spinning mode of order 2*

## Conclusions

- ❑ **Validation** of the **hybrid methodology & tools** used for this kind of **engine applications**
- ❑ **Important shielding effect** on the **2D propagation** in a **medium at rest**
- ❑ **Important** (highly sheared) **mean flow refraction effect** on the **2D propagation**
- ❑ **Significant attenuation** of the **2D shielding effect** by the **(inhomogeneous) mean flow**
- ❑ **Extension** of the **isolated nozzle** study to **3D** : still **important mean flow refraction effects**

## Perspectives

- ❑ **3D computations** over the **shielded nozzle (complex 3D geometry)**
- ❑ **Generation of exact fan noise modes**, in order to perform a **more realistic calculation (European Project “NACRE”)**